

5.2 SIMPLIFY NOTATION, LET $\gamma = \frac{z}{3(1+w)}$

$$(5.51) \quad 1+z = \left(\frac{t_0}{t_e}\right)^\gamma$$

$$(5.48) \quad t_0 = \gamma/H_0$$

$$\frac{dt_e}{dt_0} = (1+z)^{-1/\gamma}$$



$$\Rightarrow \frac{dz}{dt_0} = \gamma t_0^{\gamma-1} t_e^{-\gamma} - \gamma t_e^{-\gamma-1} t_0^\gamma \frac{dt_e}{dt_0}$$

$$= H_0 \left(\frac{t_0}{t_e}\right)^\gamma - \gamma \left(\frac{1}{t_e}\right) \left(\frac{t_0}{t_e}\right)^\gamma (1+z)^{-1/\gamma}$$

$$(5.52) \quad t_e = \frac{\gamma}{H_0} (1+z)^{-1/\gamma}$$

$$= H_0 (1+z) - \gamma \left\{ \frac{H_0}{\gamma} (1+z)^{1/\gamma} \right\} (1+z) (1+z)^{-1/\gamma}$$

$$= H_0 (1+z) - H_0 (1+z) = 0 ?$$

This would work if

$$t_e = \frac{\gamma}{H_0} (1+z)^{-1}$$

BUT THIS DIFFERS FROM
Eq. (5.52) !!

(I'll work into this.
- flip)

!!

SORRY, I COULDN'T FIGURE THIS ONE OUT!

SUPPOSE, AS PUDER SAYS,

$$\frac{dz}{dt_0} = H_0 (1+z) - H_0 (1+z)^{3(1+w)/2}$$

Then REDSHIFT DECREASES w/ TIME FOR

$$3(1+w)/2 > 1$$

$$3 + 3w > 2$$

$$\boxed{w > -\frac{1}{3}}$$

INCREASES w/ TIME FOR

$$\boxed{w < -\frac{1}{3}}$$

5.3

$$(5.81) \quad \frac{dz}{dt_0} = H_0(1+z) - H_0(1+z)^{\frac{3(1+w)}{2}}$$

MATTER ONLY $\Rightarrow w=0$

formally, we want

CHOOSE SIGN OF 10^{-6}
s.t. $\Delta t_0 > 0$

$$\Delta t_0 = \frac{1}{H_0} \int_1^{1+10^{-6}} \frac{dz}{(1+z) - (1+z)^{\frac{3}{2}}}$$

but we can just approximate this as

$$\Delta t_0 \approx \left(\frac{dz}{dt_0}\right)_{z=1} \Delta z$$

$$= 5.31 \times 10^{11} \text{ sec}$$

$$= 16,800 \text{ years}$$

(That's a long time for a small redshift!)

5.4RADIATION ONLY :

$$(5.65) \quad d_p(t_0) = \frac{c}{H_0} \frac{z}{1+z}$$

$$(5.66) \quad d_p(t_0) = \frac{c}{H_0} \frac{z}{(1+z)^2}$$

MATTER ONLY :

$$(5.60) \quad d_p(t_0) = \frac{2c}{H_0} \left[1 - \frac{1}{\sqrt{1+z}} \right]$$

$$(5.61) \quad d_p(t_0) = \frac{2c}{H_0(1+z)} \left[1 - \frac{1}{\sqrt{1+z}} \right]$$

 Λ ONLY :

$$(5.79) \quad d_p(t_0) = \frac{c}{H_0} z$$

$$(5.80) \quad d_p(t_0) = \frac{c}{H_0} \frac{z}{1+z}$$

6.1

MATTER ONLY, $\Omega_0 > 1$, $K = +1$

$$(6.16) \quad H.o.t = \int_0^\theta \frac{da}{\sqrt{\sin a + (1 - \Omega_0)}}$$

 $a_0 = 1$ BY OUR CHOICE OF NORMALIZATION

SOLUTION IS PARAMETERIZED BY (6.17) & (6.18)

$$(1) \quad a(\theta) = \frac{1}{2} \frac{\Omega_0}{\Omega_0 - 1} (1 - \cos \theta)$$

$$(2) \quad t(\theta) = \frac{1}{2H_0} \frac{\Omega_0}{(\Omega_0 - 1)^{3/2}} (\theta - \sin \theta)$$

$$\theta \in (0, 2\pi)$$

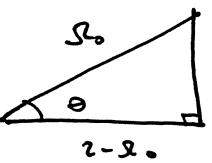
USE (1) TO DETERMINE θ

$$a_0 = 1 = \frac{1}{2} \frac{\Omega_0}{\Omega_0 - 1} (1 - \cos \theta)$$

$$\Rightarrow \frac{\Omega_0}{2(\Omega_0 - 1)} \cos \theta = \frac{\Omega_0}{2(\Omega_0 - 1)} - 1$$

$$\cos \theta = 1 - \frac{2\Omega_0 - 2}{\Omega_0}$$

$$\boxed{\cos \theta = \frac{2 - \Omega_0}{\Omega_0}}$$



$$\Leftrightarrow \sqrt{\Omega_0^2 - (2 - \Omega_0)^2} = \sqrt{4\Omega_0 - 4}$$

DETERMINE $\sin \theta$

$$\sin \theta = \frac{1}{\Omega_0} \cdot 2 \sqrt{\Omega_0 - 1}$$

PLUG INTO (2)

$$H.o.t_0 = \frac{1}{2} \frac{\Omega_0}{(\Omega_0 - 1)^{3/2}} \left\{ \cos^{-1} \left(\frac{2 - \Omega_0}{\Omega_0} \right) - \frac{2}{\Omega_0} \sqrt{\Omega_0 - 1} \right\}$$

$$\boxed{H.o.t_0 = \frac{\Omega_0}{2(\Omega_0 - 1)^{3/2}} \cos^{-1} \left(\frac{2 - \Omega_0}{\Omega_0} \right) - \frac{1}{\Omega_0 - 1}}$$

$$6.3 \quad (5.9) \quad \Sigma_w(a) = \Sigma_{w,0} a^{-3(1+w)}$$

$$\Omega_w = \frac{\Sigma_w(a)}{\Sigma_c}$$

EQUATING WHEN $l = \frac{\Omega_Q}{\Omega_M}$

$$\begin{aligned} l &= \frac{\Omega_Q}{\Omega_M} \\ &= \frac{\Omega_{Q,0} a^{-3/2}}{\Omega_{M,0} a^{-3}} \\ &= \frac{\Omega_{Q,0}}{\Omega_{M,0}} a^{3/2} \end{aligned}$$

i.e. $a_{MQ} = \left(\frac{\Omega_{M,0}}{\Omega_{Q,0}}\right)^{2/3}$

$$a_{MQ} = \left(\frac{\Omega_{M,0}}{1 - \Omega_{M,0}}\right)^{2/3}$$

THE FRIEDMANN EQUATION (6.6) TAKES THE FORM

$$\frac{H^2}{H_0^2} = \frac{\Omega_{M,0}}{a^3} + \frac{\Omega_{Q,0}}{a^{3/2}}$$

FOLLOWING THE STEPS ON P. 83

$$\frac{\dot{a}^2}{H_0^2} = \frac{\Omega_{M,0}}{a} + \Omega_{Q,0} a^{1/2}$$

$$H_0 t = \int_0^a \frac{da}{\sqrt{\Omega_{M,0}/a + \Omega_{Q,0} a^{1/2}}} \quad \leftarrow \Omega_{Q,0} = 1 - \Omega_{M,0}$$

IF YOU ARE ENTERPRISING YOU CAN SOLVE THIS (PERHAPS w/ MATHEMATICA)
BUT ALL OF THE PHYSICAL THINKING WAS UP TO HERE.

FOR $a \ll a_{MQ}$, $(\Omega_{M,0}/a) \gg (\Omega_{Q,0} a^{1/2})$

$$\Rightarrow H_0 t \approx \int_0^a da \sqrt{\frac{a}{\Omega_{M,0}}} = \frac{2}{3} (\Omega_{M,0})^{-1/2} a^{3/2}$$

$$a(t) \approx \left(\frac{9}{4} H_0^2 \Omega_{M,0} t^2\right)^{1/3}$$

FOR $a \gg a_{MQ}$, $(\Omega_{Q,0} a^{1/2}) \gg (\Omega_{M,0}/a)$

$$\Rightarrow H_0 t \approx \int_0^a da (\Omega_{Q,0} a^{1/2})^{-1/2} = \frac{4}{3} (1 - \Omega_{M,0})^{-1/2} a^{3/4}$$

$$a(t) \approx \left(\frac{3}{4} H_0 (1 - \Omega_{M,0})^{1/2} t\right)^{4/3}$$

6.4

STATIC UNIVERSE: $\dot{a} = 0, \ddot{a} = 0$

$$\text{ACCELERATION EQUATION (4.44)} \quad \frac{\ddot{a}}{a} = -\frac{4\pi G}{3c^2} (\epsilon + 3P)$$

$$\text{EQUATION OF STATE (4.50)} \quad P = w\epsilon$$

WANT Q TO BE REPULSIVE $\Rightarrow \ddot{a} > 0$
SINCE $a > 0$, THIS MEANS:

$$(\epsilon + 3P) < 0$$

$$\Rightarrow 1 + 3w_a < 0$$

$$w_a < -\frac{1}{3}$$

NOW INCLUDE MATTER AND IMPOSE $\ddot{a} = 0$

$$\Rightarrow 0 = \epsilon_Q + \epsilon_M + 3w_a \epsilon_Q$$

$$\Rightarrow \epsilon_M = -(1 + 3w_a) \epsilon_Q$$

NOW USE FRIEDMANN EQUATION (4.13) w/ $\dot{a} = 0$

$$0 = \frac{8\pi G}{3c^2} (\epsilon_M + \epsilon_Q) - \frac{Kc^2}{R_0^2} \frac{1}{a(t)^2}$$

$$K = (+\text{number}) (\epsilon_M + \epsilon_Q)
= (-+) (-3w_a \epsilon_Q)$$

$$\text{SINCE } w_a < -\frac{1}{3} \Rightarrow K \text{ IS POSITIVE}$$

$$\int \text{set } K = +1$$

$$\frac{c^2}{R_0^2 a^2} = -\frac{8\pi G}{3c^2} (3w_a \epsilon_Q) \quad | \quad a_0 = 1$$

$$R_0^2 = \frac{-c^4}{8\pi G w_a \epsilon_Q}$$

6.6

$$\begin{aligned} \Omega_0 &= \Omega_{\Lambda,0} > 1 \\ \dot{a} &> 0 \\ k &= +1 \end{aligned}$$

ASSUMPTIONS

USE eq. (6.34), FRIEDMANN EQ. FOR MATTER + CURVATURE + Λ
AND SET $\Omega_m,0 = 0$

$$\frac{H^2}{H_0^2} = \frac{1 - \Omega_{\Lambda,0}}{a^2} + \Omega_{\Lambda,0}$$

$$H = \dot{a}/a, \text{ so THERE IS A CRITICAL POINT WHEN } H=0$$

$$\Rightarrow a_{\text{crit}} = \sqrt{\frac{\Omega_0 - 1}{\Omega_0}}$$

IS THIS A "BOUNCE POINT"?

THE ACCELERATION EQUATION (4.44) TELLS US:

$$\frac{\ddot{a}}{a} = -(\text{POSITIVE } \#)(\varepsilon + 3P)$$

FOR Λ , HOWEVER, $P = -\varepsilon_\Lambda$ (4.66)

$$\Rightarrow \ddot{a} > 0$$

Thus a is CONCAVE UP, \Rightarrow SO $a_{\text{crit}} = a_{\text{bounce}}$. ✓

THE FRIEDMANN EQ GIVES US

$$H_0(t-t_{\text{bounce}}) = \int_{a_{\text{bounce}}}^a \frac{da}{\sqrt{a^2 \Omega_0 + (1-\Omega_0)}} = \frac{1}{\sqrt{\Omega_0}} \int_{a_b}^a (a^2 - a_b^2)^{-1/2} da$$

this is a tricky integral

... in particular, MATHEMATICA isn't helpful!

RECALL THAT $\frac{d}{dx}(\cosh^{-1}x) = (x^2 - 1)^{-1/2}$

SO LET'S MANIPULATE THE ABOVE INTEGRAL TO THIS FORM

$$\begin{aligned} H_0(t-t_b) &= \frac{1}{\sqrt{\Omega_0} a_b} \int_{a_b}^a ((a/a_b)^2 - 1)^{-1/2} da \quad \text{LET } b = a/a_b \\ &= \frac{a_b}{\sqrt{\Omega_0} a_b} \int_1^{a/a_b} (b^2 - 1)^{-1/2} db \\ &= \frac{1}{\sqrt{\Omega_0}} [\cosh^{-1}(a/a_b) - \cosh^{-1}(1)] \end{aligned}$$

$$\Rightarrow \cosh(a/a_b) = \sqrt{\Omega_0} H_0(t-t_b)$$

$$a(t) = a_{\text{bounce}} \cosh(\sqrt{\Omega_0} H_0(t-t_b))$$

$$t_b - t_b = \frac{1}{H_0 \sqrt{\Omega_0}} \cosh^{-1}(a(t)/a_b)$$

6.8 FROM TABLE 6.2

$$\begin{aligned}\Omega_{m,0} &= 0.3 \\ \Omega_{r,0} &= 5 \times 10^{-5} \\ \Omega_{bary,0} &= 0.04\end{aligned}$$

$$\left. \right\} \quad \Omega_i = \frac{\epsilon_i}{\epsilon_c}$$

$$(6.42) \quad d_{\text{horizon}}(t_0) = 14,000 \text{ Mpc}$$

$$\epsilon_{i,0} = \epsilon_{c,0} \Omega_{i,0} \quad \epsilon_c = \frac{3c^2}{8\pi G} H(t)^2$$

$$\Rightarrow \epsilon_{m,0} = \frac{3c^2}{8\pi G} H_0^2 \Omega_{m,0}$$

$$\begin{aligned}\text{total MASS} &= \epsilon_{m,0} \times \text{VOLUME} \\ &= \epsilon_{m,0} \times \left(\frac{4}{3}\pi (d_{\text{horizon}}(t_0))^3 \right) \\ &= \frac{3c^2}{8\pi G} H_0^2 \Omega_{m,0} \cdot \frac{4}{3}\pi d_h(t_0)^3 \\ &\approx [9 \times 10^{53} \text{ kg}]\end{aligned}$$

$$\begin{aligned}\Rightarrow \text{total RADIATION} &= \epsilon_{r,0} \times \text{VOLUME} \\ &= \frac{3c^2}{8\pi G} H_0^2 \Omega_{r,0} \cdot \frac{4}{3}\pi d_h(t_0)^3 \\ &\approx [1 \times 10^{49} \text{ J}]\end{aligned}$$

$$\Rightarrow \text{ASSUME } M_{\text{baryon}} = M_{\text{proton}}$$

$$\begin{aligned}\text{total BARYONIC MASS} &= \epsilon_{b,0} \times \text{VOLUME} \\ &= \frac{3c^2}{8\pi G} H_0^2 \Omega_{bary,0} \cdot \frac{4}{3}\pi d_h(t_0)^3\end{aligned}$$

$$\begin{aligned}\text{number of baryons} &= (\text{total baryonic mass}) / m_p \\ &\approx [7 \times 10^{79} \text{ BARYONS}]\end{aligned}$$

LAST PROBLEM

$$\begin{cases} S_{M,0} = .26 \\ S_{\Lambda,0} = .74 \\ H_0 = 70 \text{ km/s.Mpc} \end{cases}$$

a) CONDITION FOR $M-\Lambda$ EQUALITY:

$$\begin{aligned} S_{M,0} a_{M,\Lambda}^{-3} &= S_{\Lambda,0} \\ \Rightarrow a_{M,\Lambda} &= \left(\frac{S_{\Lambda,0}}{S_{M,0}} \right)^{1/3} \approx .71 \\ z_{M,\Lambda} &= a_{M,\Lambda}^{-1} - 1 = \boxed{0.42} \end{aligned}$$

b) IN THE EXPONENTIAL EXPANSION ERA, S_m IS NEGIGIBLE SO FRIEDMANN EQ:

$$\begin{aligned} \left(\frac{\dot{a}}{a} \right)^2 &= \frac{8\pi G}{3c^2} \varepsilon_\Lambda \\ \dot{a} &= \sqrt{\frac{8\pi G}{3c^2} \varepsilon_\Lambda} a(t) \end{aligned}$$

 H_0 for Λ -ONLY UNIVERSE (Eq. 5.77)

$$\Rightarrow a(t) = e^{H_0(t-t_0)}$$

$$e\text{-folding time: } \boxed{\Delta t = \frac{1}{H_0} \approx 5.1 \times 10^{12} \text{ s}}$$

$$\text{c) FRIEDMANN EQ: } H_0 t = \int_0^a \frac{da}{[S_{M,0} a^{-1} + S_{\Lambda,0} a^2]^{1/2}}$$

$$\Rightarrow dt = H_0^{-1} \{ S_{M,0} a^{-1} + S_{\Lambda,0} a^2 \}^{-1/2} da$$

$$\begin{aligned} (\text{5.35}) \quad d_{\text{horizon}}(t_0) &= c \int_0^{t_0} dt a(t)^{-1} \\ &= \frac{c}{H_0} \int_0^a \frac{da}{\sqrt{S_{M,0}/a + S_{\Lambda,0}/a^2}} \end{aligned}$$

$\max_a a \rightarrow \infty$
(can take derivative w/r t or a to confirm, or just observe)

$$\boxed{d_{\text{horiz,max}} = \frac{c}{H_0} \int_0^\infty \frac{da}{\sqrt{S_{M,0}/a + S_{\Lambda,0}/a^2}} \approx 19.8 \times 10^3 \text{ Mpc}}$$

$$\begin{aligned} d_{\text{horiz,present}} &= \frac{c}{H_0} \int_0^1 \left(\sqrt{S_{M,0}/a + S_{\Lambda,0}/a^2} \right)^{-1} da \\ &\approx 15000 \text{ Mpc} \end{aligned}$$